



Delhi Institute for Administrative Services
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ALL INDIA TEST SERIES CSE-2023

Candidate 's Information

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2. UPSC ROLL NO:- 6701477
3. MOBILE NO:-
4. SUBJECT:- Thermodynamics
5. DATE:- 12-08-23

Dias Roll No: 230001

TS-9

FOR OFFICE USE ONLY:-

Q.NO	MARKS
1.	32½
2.	22½
3.	30½
4.	
5.	26½
6.	33
7.	
8.	

TOTAL MARKS

145
250
AD

EXAMINER SIGNATURE

INVIGILATOR SIGNATURE

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Q1
(a)

Entropy is the measure of disorderliness in the system.

We know, $ds = \frac{dQ_{rev}}{T}$. Here $dQ = mC_p dT$

$$\Rightarrow 5 \times 4.18 (T - 273) = 5 \times 4.18 (353 - T)$$

$$\Rightarrow 2T = 626 \Rightarrow \boxed{T = 313}$$

Where $T =$ final temp. of water.

$$\Rightarrow ds_1 = \int_{273}^{313} mC_p \frac{dT}{T} = 5 \times 4.18 \times \ln \frac{313}{273} = 2.85 \text{ kJ/k}$$

$$ds_2 = \int_{353}^{313} mC_p \frac{dT}{T} = 5 \times 4.18 \times \ln \frac{313}{353} = -2.51 \text{ kJ/k}$$

$$\Rightarrow \text{Total } \Delta S = \Delta S_1 + \Delta S_2 = \boxed{0.336 \text{ kJ/k}}$$

\therefore Total change in entropy ~~is~~ 0.336

System is $\boxed{0.336 \text{ kJ/k}}$ i.e. $\Delta S > 0 \rightarrow$ consistent

with law of constant entropy increase.

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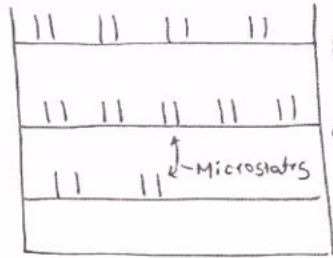
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Q1
 (b)

Maxwell-Boltzmann gives the statistical distribution for distinguishable particles.



$E_3 = 2kT$ Given $T = 100K$,

$\Rightarrow kT = 1.38 \times 10^{-21} J$.

We know, $\frac{n_i}{g_i} = A e^{-E_i/kT}$

$\Rightarrow n_1 = 2A$; $n_2 = 6A e^{-1}$
 $\& n_3 = 4A e^{-2}$

Fig 1. Macro-micro states

\Rightarrow Total $N = \sum n_i = 2A + \frac{6A}{e} + \frac{4A}{e^2} = \underline{4.38A}$

$\Rightarrow P(E_3) = \frac{n_3}{N} = \frac{4A}{e^2 \cdot 4.38A} = \boxed{12.35\%}$

$P(E_1) = \frac{n_1}{N} = \frac{2A}{4.38A} = \boxed{45.6\%}$

$\frac{6}{10}$ This verifies the distribution that at lower temperature, most of the particles are in ground state.

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Q1
 (c)

Carnot engine is a reversible engine working between 2 reservoirs of heat.

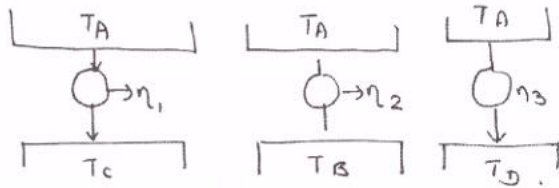


Fig 1. Carnot engine b/w reservoirs

We know,

$$\eta_{rev} = 1 - \frac{T_2}{T_1}$$

where $T_2 = \text{sink}$

$T_1 = \text{source}$

temperature.

$$\Rightarrow \eta_1 = 1 - \frac{T_C}{T_A} ; \eta_2 = 1 - \frac{T_B}{T_A} ; \eta_3 = 1 - \frac{T_D}{T_A}$$

Given $\eta_1 = \frac{1}{2} (\eta_2 + \eta_3)$

$$\Rightarrow 1 - \frac{T_C}{T_A} = \frac{1}{2} \left(1 - \frac{T_B}{T_A} \right) + \frac{1}{2} \left(1 - \frac{T_D}{T_A} \right)$$

$$\Rightarrow \frac{T_C}{T_A} = \frac{1}{2} \left(\frac{T_B + T_D}{T_A} \right)$$

$$\Rightarrow \boxed{T_C = \frac{T_B + T_D}{2}}$$

∴ Temperature of reservoir

C is mean of temperature of reservoirs

B & D

6/10



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Q)
(d)

Specific heat of material which varies with temperature ~~temperature~~ is measured at saturation.

$$> \text{We know, } s_2 - s_1 = \frac{L}{T} \Rightarrow T \frac{ds_2}{dT} - \frac{T ds_1}{T} = \frac{dL}{dT} - \frac{L}{T}$$

$$\Rightarrow \frac{dL}{dT} - \frac{L}{T} = (C_2)_{\text{sat}} - (C_1)_{\text{sat}} \text{ where}$$

$(C_2)_{\text{sat}}$ = sp. heat capacity of saturated vapor.

$(C_1)_{\text{sat}}$ = sp. heat capacity of saturated liquid.

Given: $L = 800 - 0.705T \Rightarrow \frac{dL}{dT} = -0.705$

At $T = 373 \text{ K}$, we have $\frac{L}{T} = \frac{800}{373} - 0.705 = 1.44$

$$\Rightarrow -0.705 - 1.44 = (C_2)_{\text{sat}} - (C_1)_{\text{sat}}$$

Assuming $(C_1)_{\text{sat}} = 1 \text{ cal/g-K}$

$$(C_2)_{\text{sat}} = -1.14 \text{ cal/g-K}$$

$\frac{6 \frac{1}{2}}{10}$

As $(C_2)_{\text{sat}}$ is negative, net heat has to be extracted out of the system to increase heat while keeping vapour saturated.

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Q1
(c)

Entropy is the measure of disorderliness in the system.

We know, $ds = \frac{dQ_{rev}}{T}$. For ideal gas,

$$Tds = dQ = C_v dT + Pdv.$$

Also, $Pv = RT$ for ideal gas

$$\Rightarrow Pdv + v dP = R dT.$$

$$\Rightarrow Tds = \frac{C_v}{R} [Pdv + v dP] + Pdv.$$

$$\Rightarrow Tds = C_p \frac{Pdv}{R} + C_v \frac{v dP}{R}$$

$$\Rightarrow Tds = C_p \frac{dv}{v} + C_v \frac{dP}{P}$$

$$\Rightarrow \Delta S = \int_{v_1}^{v_2} C_p \frac{dv}{v} + \int_{P_1}^{P_2} C_v \frac{dP}{P}$$

$$\Rightarrow \Delta S = C_p \ln \frac{v_2}{v_1} + C_v \ln \frac{P_2}{P_1}$$

$\frac{6 \frac{1}{2}}{10}$

Entropy can be further used to check feasibility of a reaction



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Q.2
(a)

Adiabatic demagnetization is a method of producing near zero temperatures ($\sim 0.0025\text{K}$) by isothermal magnetization - adiabatic demagnetization repeatedly of paramagnetic salt.

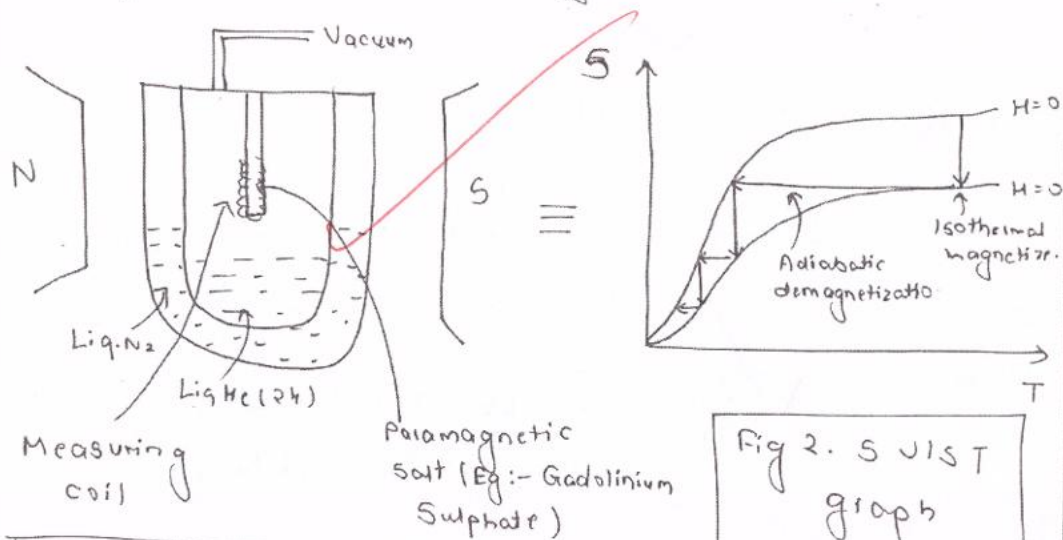


Fig. 1. Experiment setup

Fig. 2. S vs T graph

Producing low temperatures :-

- > Rock salt taken $\Rightarrow PdV = 0$
- > Isothermally magnetized \Rightarrow Dipoles aligned along magnetic field. Any excess heat is taken out by vacuum pump.
- > Adiabatically demagnetized \Rightarrow Dipoles re-align at cost of internal energy

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⇒ temperature of sample drops.

> The TdS equation for paramagnetic salt is

$$-T dS = C_H dT - \frac{C_H}{T} dH \quad \left. \begin{array}{l} C_H = \text{sp. heat capacity.} \\ C = \text{Curie's constant} \end{array} \right\}$$

During adiabatic demagnetization, $dS = 0$.

$$\Rightarrow C_H dT = \frac{C_H}{T} dH \Rightarrow \int_{T_1}^{T_2} dT(T) = \int_H^0 \frac{C}{C_H} H dH$$

$$\Rightarrow (T_2^2 - T_1^2) = \frac{C_H^2}{C_H} \quad \left. \begin{array}{l} \text{Let } T_2 - T_1 = \Delta T, \\ T_1 + T_2 = 2T_{\text{avg}}. \end{array} \right\}$$

$$\Rightarrow \boxed{\Delta T = \frac{C_H^2}{2 C_H T_{\text{avg}}}}$$

> Given, C_H (for 0.1 gm) = $1 \times 10^{-3} \text{ cal/k}$

$$\Rightarrow \Delta T = \frac{0.05 (10^8) \times 10^{-7} \times 0.1}{2 \times 10^{-3} \times 3} \quad \left. \begin{array}{l} \text{As } C = \text{erg kg}^{-1} \text{oe}^{-2} \\ \text{in CGS.} \end{array} \right\}$$

$$\Rightarrow \boxed{\Delta T = 1.99 \text{ K}} \quad \text{Here } C_H \text{ considered in Joules.}$$

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As can be seen, the drop is significant due to pre-cooling at 3k.

Applications of adiabatic demagnetization

- ① In-cryogenics of rockets
- ② Producing super-conductors.

However, scalability, pre-cooling remain the limitations

13/20

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Q2
(b)

Carnot engine is a reversible engine of 2 isothermal, 2 adiabatic process operating between 2 temperatures.

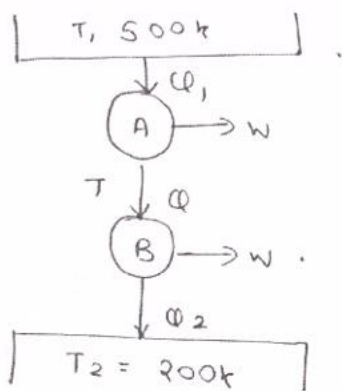


Fig 1. Carnot engine in series

> We know, for Carnot engine,

$$\eta = 1 - \frac{T_2}{T_1} = 1 - \frac{Q_2}{Q_1} = \frac{W}{Q_1}$$

where Q_1 = heat absorbed.

Q_2 = heat rejected.

T_2 = sink temperature

T_1 = source temperature

$$\Rightarrow 1 - \frac{T}{500} = \frac{W}{Q_1} = 1 - \frac{Q}{Q_1} \quad \text{--- (1)}$$

$$1 - \frac{200}{T} = \frac{W}{(Q/2)} = 1 - \frac{Q_2}{(Q/2)} \quad \text{--- (2)}$$

$$\text{From eq. (1), } \frac{Q}{Q_1} = \frac{T}{500} \quad \text{--- (3)}$$

$$\text{Also, } \left(1 - \frac{T}{500}\right) Q_1 = \left(1 - \frac{200}{T}\right) \frac{Q}{2} = W$$

$$\Rightarrow \left(1 - \frac{T}{500}\right) = \left(1 - \frac{200}{T}\right) \frac{1}{2} \left(\frac{T}{500}\right) \quad \left. \vphantom{\Rightarrow} \right\} \text{From eq. (3)}$$

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$$\Rightarrow 1 - \frac{T}{500} = \frac{T}{1000} - \frac{1}{5}$$

$$\Rightarrow \frac{4}{5} = \frac{3T}{1000} \Rightarrow T = 266.67k$$

\therefore The first engine rejects heat at $266.67k$.

Carnot engine is the engine with maximum efficiency operating between two temperatures.

02/15



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Q 2
(c)

Maxwell's equations gives the relation
between pressure, temperature, entropy, volume

We know $\frac{\partial P}{\partial T} =$ Pressure coeff. of
expansion.

$$\Rightarrow \left. \frac{\left(\frac{\partial P}{\partial T}\right)_V}{\left(\frac{\partial P}{\partial T}\right)_S} \right\} \begin{array}{l} \text{As } P, T, S \text{ are state} \\ \text{functions,} \\ F(P, T, S) = 0 \end{array}$$

$$\Rightarrow \left(\frac{\partial P}{\partial T}\right)_S \left(\frac{\partial T}{\partial S}\right)_P \left(\frac{\partial S}{\partial P}\right)_T = -1$$

$$\Rightarrow \left(\frac{\partial P}{\partial T}\right)_S = - \left(\frac{\partial S}{\partial T}\right)_P \left(\frac{\partial P}{\partial S}\right)_T \quad \text{--- (1)}$$

Show!

$$\text{We know, } C_p - C_v = T \left(\frac{\partial V}{\partial T}\right)_P \left(\frac{\partial P}{\partial T}\right)_V \quad \text{--- (2)}$$

Substituting values from eq. (1) & (2),

$$\frac{(C_p - C_v)}{T \left(\frac{\partial V}{\partial T}\right)_P \cdot \left(-\frac{\partial P}{\partial S}\right)_T \left(\frac{\partial S}{\partial T}\right)_P}$$

$$\text{We know, } T \left(\frac{\partial S}{\partial T}\right)_P = \left(\frac{\partial Q}{\partial T}\right)_P = C_p$$

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From Maxwell's equations, $\left(\frac{\partial v}{\partial T}\right)_P = -\left(\frac{\partial s}{\partial P}\right)_T$

$$\Rightarrow \left(\frac{\partial v}{\partial T}\right)_P \left(\frac{-\partial P}{\partial s}\right)_T = 1$$

\Rightarrow Given equation becomes :-

$$\frac{c_p - c_v}{c_p} = 1 - \frac{c_v}{c_p} = \boxed{1 - \frac{1}{\gamma}}$$

~~7/2~~ Hence, ratio of pressure coefficient of expansion at constant volume to constant entropy is $1 - \frac{1}{\gamma}$

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Q3
(a)

Diesel engine is a 4-stroke, 3 valve engine operating with diesel as a fuel.

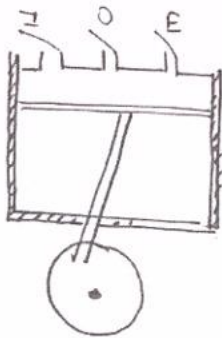


Fig 1. Diesel engine

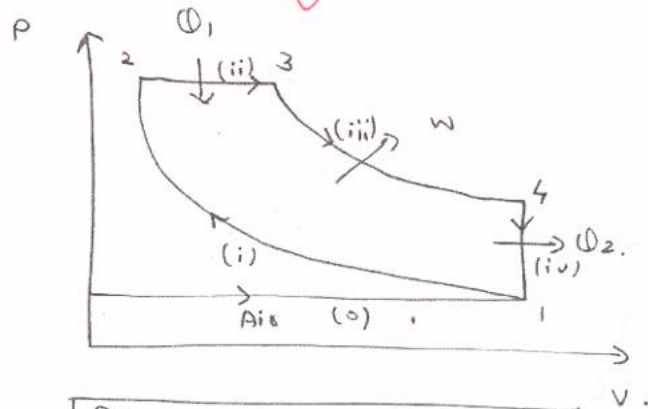


Fig 2. P-V diagram of diesel engine

Working

- > Air sucked in at constant pressure (0).
- > Stroke - I :- Adiabatic compression
Air compressed from volume v_1 to v_2 , piston moves up \rightarrow Air heated up $\therefore p v^\gamma = \text{const}$
- > Stroke - II :- Isobaric heat addition
Diesel sprinkled in heated air \rightarrow burns
- > Stroke - III :- Adiabatic work
Piston pushed down, volume expands from v_3 to v_4 $\therefore p v^\gamma = \text{const}$

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> Stroke -IV :- isochoric heat rejection

Exhaust valve opened up, heat allowed to escape.

$$\text{Efficiency } \eta = 1 - \frac{Q_2}{Q_1} = 1 - \frac{C_V (T_4 - T_1)}{C_P (T_3 - T_2)}$$

As process (i) & (iii) are adiabats,

$$P_1 V_1^\gamma = P_2 V_2^\gamma \quad \& \quad P_3 V_3^\gamma = P_4 V_4^\gamma \quad \left. \begin{array}{l} \text{Assuming} \\ \text{air} \\ \text{ideal} \\ \text{gas.} \end{array} \right\}$$

$$\Rightarrow T_1 V_1^{\gamma-1} = T_2 V_2^{\gamma-1} \quad \& \quad T_3 V_3^{\gamma-1} = T_4 V_4^{\gamma-1}$$

Writing all temperatures in terms of T_3 ,

$$T_4 = T_3 \left(\frac{1}{P_4} \right)^{\frac{\gamma-1}{\gamma}} \quad \text{--- (1)}$$

$$\text{Also, } V_1 = V_4 \Rightarrow \frac{T_3}{T_4} V_3^{\gamma-1} = \frac{T_2}{T_1} \left(\frac{V_2}{V_3} \right)^{\gamma-1}$$

$$\Rightarrow T_2 = \frac{T_3}{T_4} \cdot T_1 \left(\frac{V_3}{V_2} \right)^{\gamma-1} \neq T_1$$

$$\Rightarrow \eta = 1 - \frac{1}{\gamma} \left[\frac{\left(\frac{T_3}{P_4} \right)^{\frac{\gamma-1}{\gamma}} - T_1}{T_3 - T_2} \right]$$

$$\text{Also, } \frac{T_1 V_1^{\gamma-1}}{T_4 V_4^{\gamma-1}} = \frac{T_2 V_2^{\gamma-1}}{T_3 V_3^{\gamma-1}} \Rightarrow \frac{T_1}{T_4} = \frac{T_2}{T_3} \left(\frac{V_2}{V_3} \right)^{\gamma-1}$$

$$\text{as } V_1 = V_4$$

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$$\eta = 1 - \frac{1}{\gamma} \frac{T_4 \left[1 - \frac{T_1}{T_4} \right]}{T_3 \left[1 - \frac{T_2}{T_3} \right]} = 1 - \frac{1}{\gamma} \frac{T_4}{T_3} \frac{\left[1 - \left(\frac{v_2}{v_3} \right)^{\gamma-1} \right]}{\left(1 - \frac{T_2}{T_3} \right)}$$

Now, $\frac{T_4}{T_3} = \left(\frac{1}{\rho_c} \right)^{\gamma-1}$; $\frac{v_2}{v_3} = \frac{\rho_c}{\rho_c}$.

Substituting, we get,

$$\eta = 1 - \frac{1}{\gamma} \frac{\left(\frac{1}{\rho_c} \right)^{\gamma} - \left(\frac{1}{\rho_c} \right)^{\gamma}}{\left(\frac{1}{\rho_c} \right) - \left(\frac{1}{\rho_c} \right)}$$

For Carnot engine, $\eta = 1 - \frac{T_{\min}}{T_{\max}} = 1 - \frac{T_1}{T_3}$.

$$\eta_{\text{diesel}} = 1 - \frac{1}{\gamma} \frac{[T_1] \left[\frac{T_4}{T_1} - 1 \right]}{T_3 \left[1 - \frac{T_2}{T_3} \right]}$$

Now as $\frac{T_4}{T_1} > 1$ & $\frac{T_2}{T_3} < 1$, $\left(\frac{\frac{T_4-1}{T_1}}{1 - \frac{T_2}{T_3}} \right) > 1$.

$\Rightarrow 1 - \left(\frac{T_1}{T_3} \right) (A) \left(\frac{1}{\gamma} \right) = \eta_{\text{diesel}}$ as $(A) > 1$,

$$1 - \frac{T_1}{T_3} (A) < 1 - \frac{T_1}{T_3} \Rightarrow \eta_{\text{diesel}} < \eta_{\text{Carnot}}$$

Typically $\rho_c \approx 5$; $\rho_c \approx 17$

by design $\Rightarrow \eta_{\text{diesel}}$ is $\sim 56\%$ for

cars

3/22

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Q3
(b)

Maxwell's relations are thermodynamic relations which relate P, S, T, V

∴ We have to find $\frac{\Delta Q}{\Delta P}$ } as pressure is increased on water.

$$\Rightarrow \left(\frac{\partial Q}{\partial P}\right)_T = T \left(\frac{\partial S}{\partial P}\right)_T \quad \text{We know from Maxwell's relations,}$$

$$- \left(\frac{\partial S}{\partial P}\right)_T = \left(\frac{\partial V}{\partial T}\right)_P$$

$$\Rightarrow \left(\frac{\partial Q}{\partial P}\right)_T = -T \left(\frac{\partial V}{\partial T}\right)_P \quad \left. \begin{array}{l} \text{We know, } \alpha = \frac{1}{V} \frac{\partial V}{\partial T} \end{array} \right\}$$

$$\Rightarrow \Delta Q = -T \alpha (V) \Delta P$$

$$\text{Volume of } 10 \text{ gm water} = 10 \text{ cc}^3 = 10 \times 10^{-6} \text{ m}^3$$

$$\Rightarrow \Delta Q = -273 (-6.7 \times 10^{-5}) (10 \times 10^{-6}) (1000 \times 10^5)$$

$$\Rightarrow \Delta Q = 18.29 \text{ J} = 1.829 \text{ J}$$

As ΔQ is positive, heat is absorbed by the system & is equal to 18.3 J .

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Q.3
(c)

Clausius Clapeyron equation gives variation of boiling point with pressure.

We know, $\left(\frac{\partial P}{\partial T}\right) = \frac{L}{T(v_2 - v_1)}$ } $L = \text{latent heat,}$
 $v = \text{specific volume}$
 $T = \text{boiling point}$

Assuming for vapour stage, $v_2 \gg v_1$,

$\frac{\partial P}{\partial T} = \frac{L}{Tv_2}$ } Assuming vapour behaves like ideal gas,

$\frac{\partial P}{\partial T} = \frac{L}{T\left(\frac{RT}{P}\right)} \Rightarrow \frac{dP}{P} = \frac{L}{R} \frac{dT}{T^2}$

$\Rightarrow \int_{P_1}^{P_2} \frac{dP}{P} = \frac{L}{R} \int_{T_1}^{T_2} \frac{dT}{T^2} \Rightarrow \ln \frac{P_2}{P_1} = -\frac{L}{R} \left[\frac{1}{T_2} - \frac{1}{T_1}\right]$

Here $P_2 = 1 \text{ mm Hg}$ $T_2 = 1.2 \text{ K}$

$P_1 = 760 \text{ mm Hg (atm)}$ $T_1 = 4.2 \text{ K}$

$\Rightarrow \ln \left(\frac{1}{760}\right) = -\frac{L}{R} \left[\frac{1}{1.2} - \frac{1}{4.2}\right]$

$\Rightarrow L = 11.14R \Rightarrow L = 92.65 \text{ J/mol-K}$

This phenomenon is used in pressure cooker where pressure is increased to increase boiling point of water.

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0.5
(a)

Mean free path (λ) is the average distance traveled by molecule before it encounters another molecule.

> Coefficient of viscosity (η) = Force per unit area per unit velocity gradient.

$$\text{We know, } \eta = \frac{1}{3} \rho \bar{v} \lambda$$

where \bar{v} = average velocity = $\sqrt{\frac{8kT}{\pi m}}$ } Assuming Maxwell Boltzmann statistics.

At NTP, $T = 20^\circ \text{C} = 293 \text{K}$, $m_{\text{N}_2} = 28 \text{amu}$.

$$\Rightarrow \bar{v} = 469 \text{ m/sec}$$

$$\Rightarrow \lambda = \frac{3 \times 1.7 \times 10^{-4} \times 10^{-3}}{1.2} \times \frac{1}{469}$$

$$\Rightarrow \lambda = 9.05 \times 10^{-6} \text{ m.}$$

\therefore Mean free path of N_2

molecule is $\boxed{9.05 \mu\text{m}}$. This can be further used to estimate radius of nitrogen.



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Q5
(b)

Hydrogen at NTP can be considered as set of indistinguishable particles i.e. it follows Maxwell Boltzmann statistics.

$$\text{We know, } n(\epsilon)d\epsilon = A e^{-\epsilon/kT} \cdot g(\epsilon)d\epsilon$$

$$\text{where } g(\epsilon) = \frac{2\pi V}{h^3} (2m)^{3/2} \cdot \epsilon^{1/2}$$

$$A = \frac{N}{Z} \quad \text{where } Z = \frac{V}{h^3} (2\pi mkT)^{3/2} \quad \left. \vphantom{Z} \right\} \text{Partition Function.}$$

$$\Rightarrow n(\epsilon) = \frac{2N}{\sqrt{\pi} (kT)^{3/2}} \cdot \epsilon^{1/2} \cdot e^{-\epsilon/kT} d\epsilon$$

At NTP, $T = 293\text{K}$ is assuming $\epsilon = 0.0235\text{eV}$,

$$n(\epsilon) = \frac{2N}{\sqrt{\pi} (2.5 \times 10^{-31})} \times (6.14 \times 10^{-11}) \times (0.394) \times 1.6 \times 10^{-23}$$

$$\Rightarrow \frac{n(\epsilon)}{N} = 1.75 \times 10^{-3}$$

At NTP, 22.4 litres has N_A no. of molecules

$$\Rightarrow N \text{ for litre} = \frac{N_A}{22.4}$$

$$\Rightarrow n(\epsilon) = 4.69 \times 10^{19}$$

$$\therefore \sim 4.7 \times 10^{19} \text{ molecules would}$$

be between the given energy range.

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Q5
 (C)

Maxwell-Boltzmann statistics gives the distribution for distinguishable particles

$$\frac{n_i}{g_i} = f(\epsilon_i) = A e^{-\epsilon_i/kT}$$

$n_i = \# \text{ particles}$
 $g_i = \text{density of states}$
 $= 1 \text{ assumed here.}$

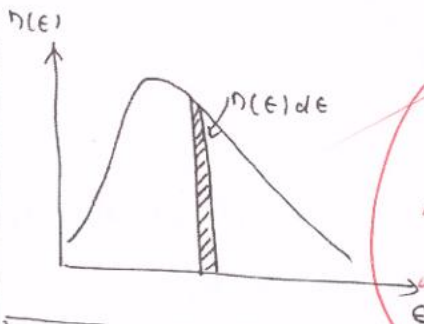
Energy levels are $-2kT, 0, 2kT$

$$\Rightarrow n_1 = A e^2 ; n_2 = A ; n_3 = A/e^2$$

$$\Rightarrow \text{Total no. of particles} = N = \sum n_i = A \left[e^2 + 1 + \frac{1}{e^2} \right] = 8.52A$$

Probability of finding the system in

$$E=0 \text{ state } P(\epsilon_2) = \frac{n_2}{N}$$



$6A \Rightarrow P(\epsilon_2) = \frac{A}{8.52A} = 0.117$

Probability of finding particle in $E=0$ state is 11.7.1.

Fig.1. Maxwell Boltzmann distribution

This is consistent with theory that larger no. of particles occupy lower levels.

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में प्रश्न संख्या के
अतिरिक्त कुछ
न लिखें।

Q5
(d)

Maxwell-Boltzmann, Bose-Einstein, Fermi-Dirac
are three statistics which give distribution
of distinguishable particles, bosons (full integral
spins & fermions (half-integral spin) respectively.

① Maxwell-Boltzmann

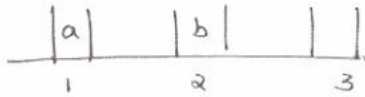


Fig 1. 3 states, 2 particles

ways in which particle can
occupy same state = 3

Total no. of ways = $3^2 = 9$

\Rightarrow # ways in which particles
occupy different state = $9 - 3 = 6$

$$\Rightarrow \frac{P(\text{same state})}{P(\text{diff. state})} = \frac{3}{6} = \boxed{\frac{1}{2}}$$

② Bose-Einstein :

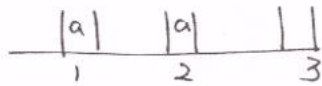


Fig 2. BE distribution

Particles are indistinguishable

\Rightarrow Total no. of ways

$$= {}^{g+n-1}C_{n-1} = {}^4C_2 = 6$$

Total no. of ways to

occupy same state = 3

$$\Rightarrow \frac{P(\text{same state})}{P(\text{diff. state})} = \frac{3}{6} = \boxed{\frac{1}{2}}$$

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③ Fermi-Dirac:- Fermions are half-spin particles by virtue of which they follow Pauli's exclusion principle \Rightarrow No two particles can have same state

$$\Rightarrow \frac{P(\text{same state})}{P(\text{diff. state})} = \boxed{0}$$

7/10

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Q5
(c)

Latent heat is the heat which is absorbed by a body at constant temperature when it changes phases.

We know, from Clausius Clapeyron equation,

$$S_2 - S_1 = \frac{L}{T} \quad \text{--- (1)}$$

$$\Rightarrow T \frac{dS_2}{dT} - T \frac{dS_1}{dT} = \frac{dL}{dT} - \frac{L}{T}$$

$$\Rightarrow \frac{dL}{dT} - \frac{L}{T} = (C_2)_{\text{sat}} - (C_1)_{\text{sat}}. \text{ So latent}$$

heat can become 0 in 2 cases.

(1) When $\Delta S = 0$ i.e. phase change occurs at constant entropy. We know $\left(\frac{\partial G}{\partial T}\right)_P = -S$

\Rightarrow If $S = 0$, $\left(\frac{\partial G}{\partial T}\right) = 0 \Rightarrow$ This will happen

For order of transition > 1

(2) When specific heat of saturated vapour = specific heat of saturated liquid & dL/dT variation is small $\Rightarrow L \approx 0$

In such cases, liquid will immediately turn into vapour as it reaches boiling point.

$\frac{dP}{dT} = \frac{L}{T(V_2 - V_1)} \left(\frac{dL}{dT}\right)$
 $L = T(V_2 - V_1) \left(\frac{dP}{dT}\right)$
 L can be zero when either $V_2 = V_1$ or $\left(\frac{dP}{dT}\right) = 0$
CRITICAL POINT

Q6
(a)

(i)

(ii)

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Q6 (a)

Real gases are the gases which deviate from ideal gas giving different eq. of state like $P(V-b) = RT$.

(i)

We know, $dQ = dU + PdV$ - 1st law of thermodynamics.

Now, $dU = \left(\frac{\partial U}{\partial V}\right)_T dV + \left(\frac{\partial U}{\partial T}\right)_V dT$. But in this gas, there is no intermolecular interaction

as a (Vanderwall constant) = 0 $\Rightarrow \frac{\partial U}{\partial V} = 0 = T\left(\frac{\partial S}{\partial V}\right)_T - P$
 $\Rightarrow dU = C_V dT$. *show* $= T\left(\frac{\partial P}{\partial T}\right)_V - P = \frac{RT}{V-b} - P = 0$
 $\Rightarrow dQ = C_V dT + PdV$. } As process is isothermal, $dT = 0$.

04

$$dW = PdV = \frac{RT}{(V-b)} dV \Rightarrow \Delta W = RT \int_{V_1}^{V_2} \frac{dV}{(V-b)}$$

$$\Rightarrow \Delta W = \Delta Q = \boxed{RT \ln \left(\frac{V_2 - b}{V_1 - b} \right)}$$

$V_2 =$ expanded volume,
 $V_1 =$ initial volume.

(ii)

In adiabatic expansion,

$$dQ = 0 \text{ \& } PV^\gamma = \text{constant} \Rightarrow \text{Here } P(V-b)^\gamma = k$$

$$\Rightarrow dW = PdV = \frac{k}{(V-b)^\gamma} dV \Rightarrow \Delta W = \int_{V_1}^{V_2} \frac{k}{(V-b)^\gamma} dV$$

$$\Rightarrow \Delta W = \frac{k}{1-\gamma} \left[\frac{V_2}{V_2^\gamma} - \frac{V_1}{V_1^\gamma} \right] = \frac{1}{(\gamma-1)} (P_1(V_1-b) - P_2(V_2-b))$$



⇒ For adiabatic, $\Delta Q = 0$

$$\oint \Delta W = \frac{1}{\gamma-1} [P_1(V_1-b) - P_2(V_2-b)]$$

(ii)

Isothermal expansion

$$\Delta W = RT \ln \left(\frac{V_2-b}{V_1-b} \right) = RT \ln \left(\frac{20.02 - 0.02}{0.2 - 0.02} \right)$$

$$\Rightarrow \Delta W = 4.71 RT = \underline{7833.1 \text{ J}}$$

In isothermal, $\Delta U = 0$, $\Delta Q = \Delta W$

$$\Rightarrow \Delta U = 0, \Delta Q = \Delta W = \underline{7.83 \text{ kJ}}$$

Adiabatic expansion

$$\Delta W = \frac{R(T_1 - T_2)}{\gamma - 1} \quad \left. \begin{array}{l} \text{where } T_1 = 200 \text{ K} \\ \text{Now, we know, } P(V-b)^\gamma = k \end{array} \right\}$$

$$\Rightarrow T(V-b)^{\gamma-1} = \text{constant}$$

$$\Rightarrow T_1(V_1-b)^{\gamma-1} = T_2(V_2-b)^{\gamma-1}$$

$$\Rightarrow 200(0.2 - 0.02)^{2/3} = T_2(20.02 - 0.02)^{2/3}$$

$$\Rightarrow \underline{T_2 = 8.65 \text{ K}}$$

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$$\Rightarrow \Delta W = \frac{R(200 - 8.65)}{\frac{5}{3} - 1} = \underline{2386.5 \text{ J}}$$

In adiabatic, $\Delta Q = 0$, $\Rightarrow \Delta U = -\Delta W$.

$$\therefore \Delta U = -2.38 \text{ kJ}, \Delta Q = 0, \Delta W = 2.38 \text{ kJ}$$

isothermal process is ideal
input process whereas, adiabatic
process is ideal output process.

~~14
20~~



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Q.6
(b)

Joule - kelvin coefficient in the relative drop in temperature with drop in pressure when a real gas is throttled through a porous plug.

$$\therefore \mu = \text{Joule kelvin coefficient} = \left(\frac{\partial T}{\partial P} \right)_H \quad \left. \vphantom{\left(\frac{\partial T}{\partial P} \right)_H} \right\} \text{As throttling is isenthalpic.}$$

$$\text{Let } H(P, T) \Rightarrow dH = \left(\frac{\partial H}{\partial P} \right)_T dP + \left(\frac{\partial H}{\partial T} \right)_P dT.$$

$$\text{As } dH = 0, \quad \& \left(\frac{\partial H}{\partial T} \right)_P = C_p \quad \& \partial H = \partial U + \partial(PV).$$

$$\Rightarrow 0 = \frac{\partial}{\partial P} (U + PV) dP + C_p dT.$$

$$\Rightarrow \mu = \left(\frac{\partial T}{\partial P} \right) = -\frac{1}{C_p} \left[\frac{\partial U}{\partial P} + \frac{\partial (PV)}{\partial P} \right]$$

> For Joule's law, $\frac{\partial U}{\partial P} = 0$ as there is no

intermolecular interaction. But for real

gases, $\frac{\partial U}{\partial P} \neq 0 \Rightarrow$ Deviates from

Joule's law.

> Boyle's law is $\frac{\partial}{\partial P} (PV) = 0$. Hence we

get $PV = RT$ for ideal gas. But for

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real gas, $\frac{\partial (pv)}{\partial P} \neq 0 \Rightarrow$ Deviation from Boyle's law.

The first term $\left(\frac{\partial U}{\partial P}\right)$ always produces cooling while $\frac{\partial (pv)}{\partial P}$ can produce both cooling/heating.

Case of H_2 & He

> Assuming H_2, He following Vander Waals eq.,

$$(P + a/v^2)(v-b) = RT \quad ; \quad \text{we get } \mu = \frac{1}{C_p} \left[\frac{2a}{RT} - b \right]$$

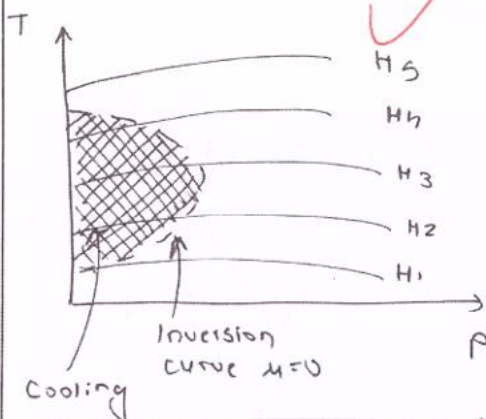


Fig. 1. T vs P for iso enthalpic process

We define inversion temp.

$$\text{as } T_i = \frac{2a}{Rb} \quad \text{IF } T < T_i,$$

only then $\mu > 0 \Rightarrow$ cooling occurs.

But for He, $T_i = -240^\circ C$

& for H_2 , $T_i = -80^\circ C$.

As normally $T > T_i$,

H_2 & He heat in Joule-Kelvin expansion.

This process is used to liquify gases in petrochemical industries via Linde's technique.

~~9.5~~
15



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Q.6
(c)

Specific heat is the heat required to increase temperature of unit mass of substance by 1 unit. Einstein & Debye gave two theories for specific heat :-

	<u>Debye</u>	<u>Einstein</u>
C_v	$C_v = 9R \left[4 \left(\frac{T}{\theta_D} \right)^3 \int_0^{\theta_D/T} \frac{x^3}{e^{x^3}-1} dx - \frac{\theta_D}{T} \frac{1}{e^{\theta_D/T}-1} \right]$	$C_v = 3R \left(\frac{\theta_E}{T} \right)^2 \frac{e^{\theta_E/T}}{(e^{\theta_E/T}-1)^2}$
	$\theta_D = \text{Debye temp.}$	where $\theta_E = \text{Einstein temp}$

Assump-tions

(1) Molecules behave like Planck oscillator vibrating with different frequencies
[$\nu_1, \nu_2, \dots, \nu_m$]

(1) Molecules vibrate as Planck oscillator, all with same frequency ν .

(2) No concept of individual degree of freedom.

(2) Each molecule has 3 degree of freedom

Total # modes = $3N$

(3) Vibration is elastic i.e. has both transverse & longitudinal component

(3) No such concept

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- Limitations
- ① Valid only for isotropic solids
 - ① Cannot explain $C_v \propto T^3$ at low temperature.
 - ② No explanation for semi-empirical curve fitting method to find Θ_D .

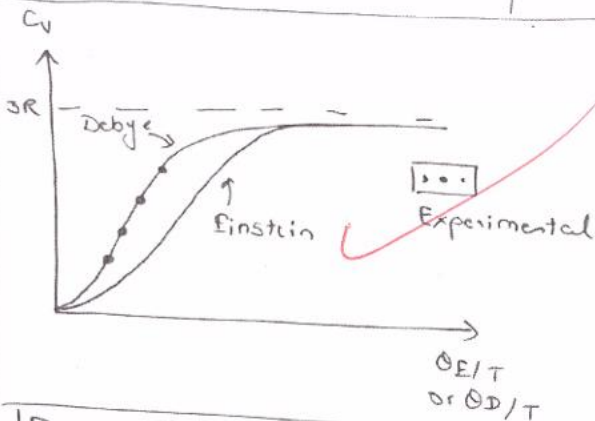


Fig 1. C_v v/s temperature

Given :-

$$U = \frac{9Nh}{v_m^3} \int_0^{v_m} \frac{v^2 dv}{e^{h\nu/kT} - 1}$$

Let $\frac{hv_m}{k} = \Theta_D$

Let $\frac{h\nu}{kT} = x$

$\Rightarrow x$ varies from 0 to Θ_D/T .

$$\Rightarrow U = \frac{9Nh}{v_m^3} \cdot \left(\frac{kT}{h}\right)^3 \int_0^{\Theta_D/T} \frac{x^2 dx}{e^x - 1}$$

$$\Rightarrow U = 9Nh \left(\frac{T}{\Theta_D}\right)^3 \int_0^{\Theta_D/T} \frac{x^2 dx}{e^x - 1}$$

} Using Leibnitz rule, $C_v = \frac{\partial U}{\partial T}$

$$C_v = 9Nh \left[3 \left(\frac{T^2}{\Theta_D^3}\right) \int_0^{\Theta_D/T} \frac{x^2 dx}{e^x - 1} + \left(\frac{T}{\Theta_D}\right)^3 \cdot \left(-\frac{\Theta_D}{T^2}\right) \left(\frac{\Theta_D}{T}\right)^2 \frac{1}{(e^{\Theta_D/T} - 1)} \right]$$

$$\Rightarrow C_v = 9Nh \left[3 \left(\frac{T^2}{\Theta_D^3}\right) \int_0^{\Theta_D/T} \frac{x^2 dx}{e^x - 1} - \frac{1}{e^{\Theta_D/T} - 1} \right]$$

At low temp. this will show $C_v \propto T^2$